

# Results from Ground and Flight Testing of UAS Weather Payload

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**Abstract:** Prior, during and post atmospheric characterization field campaigns and the data reduction associated with calculating temperature structure constant,  $C_n^2$ , from nodal sensors is both well-known and misrepresented. Critical to accurate estimates of is selecting, and what this author refers to as, a Structure Function Window Length from which to base temperature fluctuations on. To best understand the impact of the Structure Function Window (SFV), a set of data taken over 24 hours, that included co-located nodal Differential Temperature Sensor (DTS) systems, nodal miniature Anemoment sonic anemometers, a larger Applied Technologies sonic anemometer, and an integrated path Scintec BLS900 atmospheric characterization system, was used to calculate and compare a variety of moving mean lengths which ultimately build up. The SFV and the methods for estimating the length of the SFV to best estimate is the focus of this presentation.

## Refractive Index Structure Coefficient ( $C_n^2$ )

Measurement of refractive-index structure coefficient,  $C_n^2$ , by direct measurement of thermal inhomogeneities in the atmosphere. The values of  $C_n^2$  obtained from visible wavelength measurements can be directly related to  $C_T^2$ , the temperature structure coefficient. The relationship is defined as:

$$\text{Differential Temperature } C_T^2 = \frac{(\Delta T)^2}{r^{2/3}}$$

$$\text{Power Spectrum } C_T^2 = \frac{-8/3 \Phi(2\pi/U_m)}{f^{-2/3}}$$

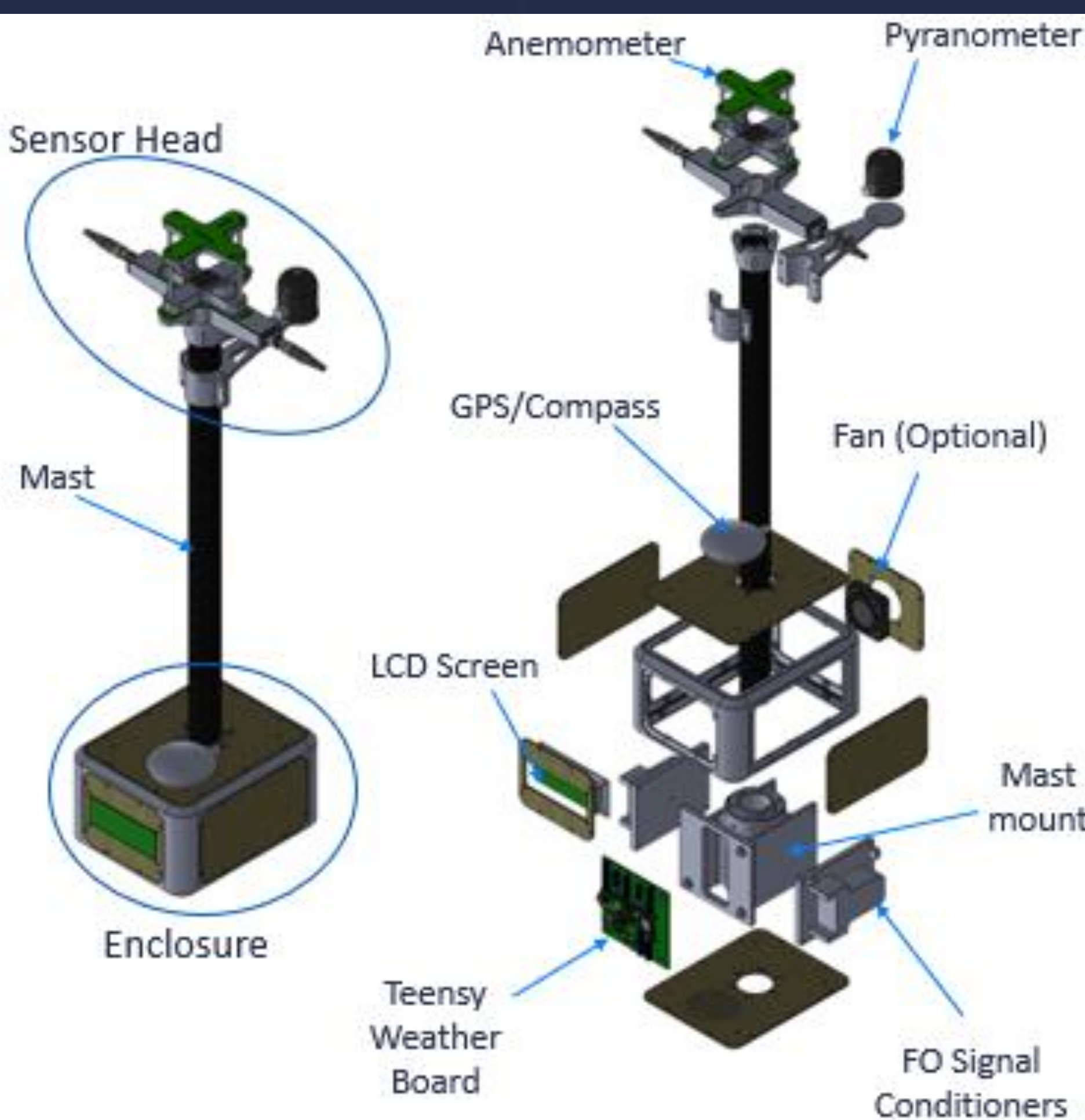
$$C_n^2 = \left[ 79 \frac{P}{T^2} \right]^{-1} \times 10^{-12} C_T^2$$

Where for differential temperature  $\Delta T$  is the ensemble average obtained from a pair of temperature sensors separated by a distance  $r$ ,  $P$  is pressure in millibars, and  $T$  is a temperature in kelvins. Using a power spectrum method  $\Phi$  is the temperature power spectrum,  $U_m$  is the mean wind speed, and  $f$  is the spectrum frequency range.



New Buoy/Sea ODS built for the Navy

## Platform Support



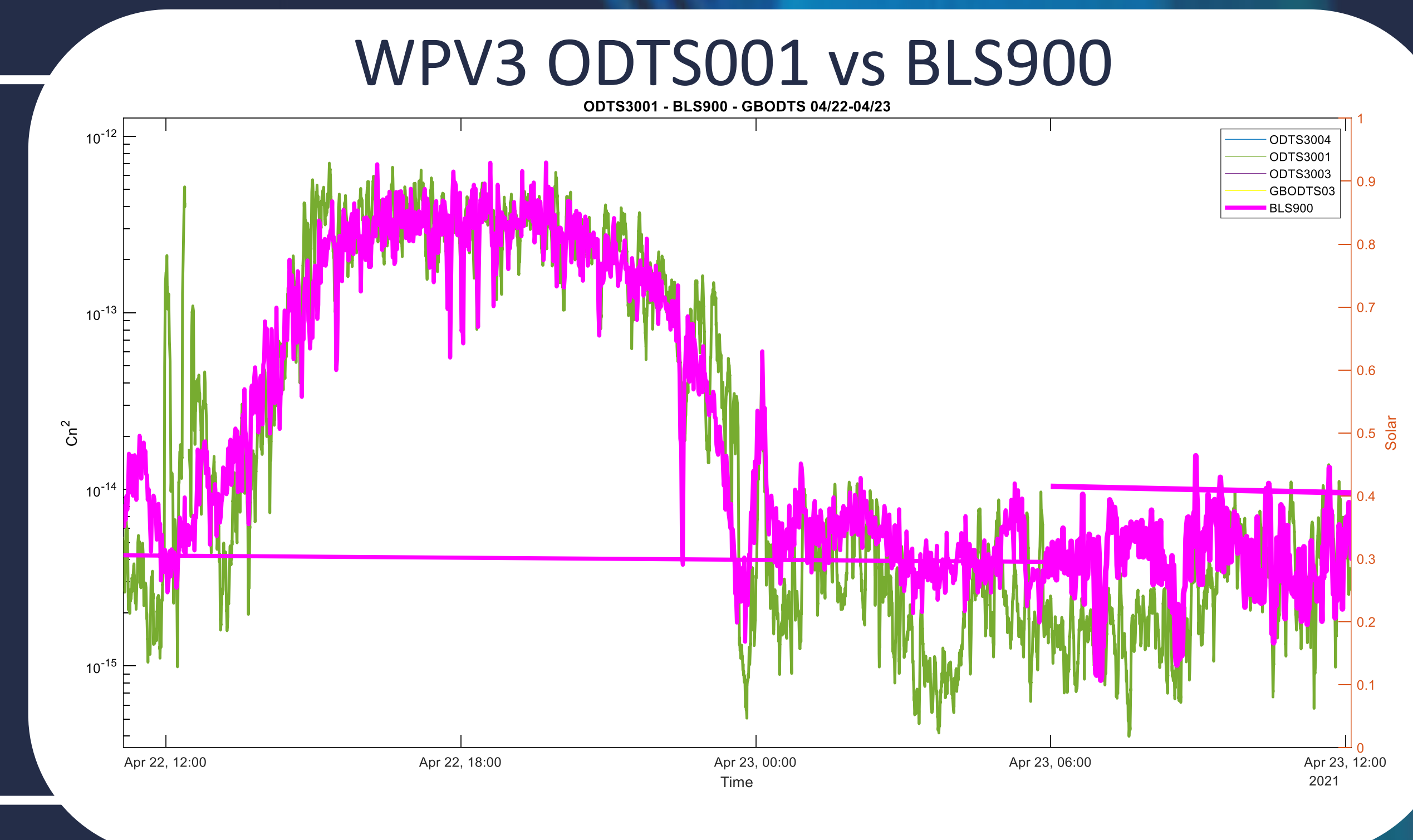
## UAS Payload System Design

### Critical Features:

- Rigid Structure
- Sensor Spacing away from Propellers
- Secure Cabling
- Field Replaceable Wear Items
- Quick Health and Status Reference
- Low Size Weight And Power (SWAP)
- On Board Data Storage (SD Card)
- GPS
- Altimeter
- Wide Power Input Range

## Ground (Validation) Calibration and Correlation Test Purpose

- Collect atmospheric turbulence data for the updated BlueHalo WPV3 Atmospheric Characterization Payloads
- Collect and compare atmospheric turbulence data from various sensors – WPV3, GBDTS, ATI Sonic Anemometer, and BLS900
- Collect atmospheric turbulence data to support comparison and correlation of ODS data vs BLS900 data. Correlate minimum Structure Function Window (SFV) with ODS data reduction using BLS900 data.

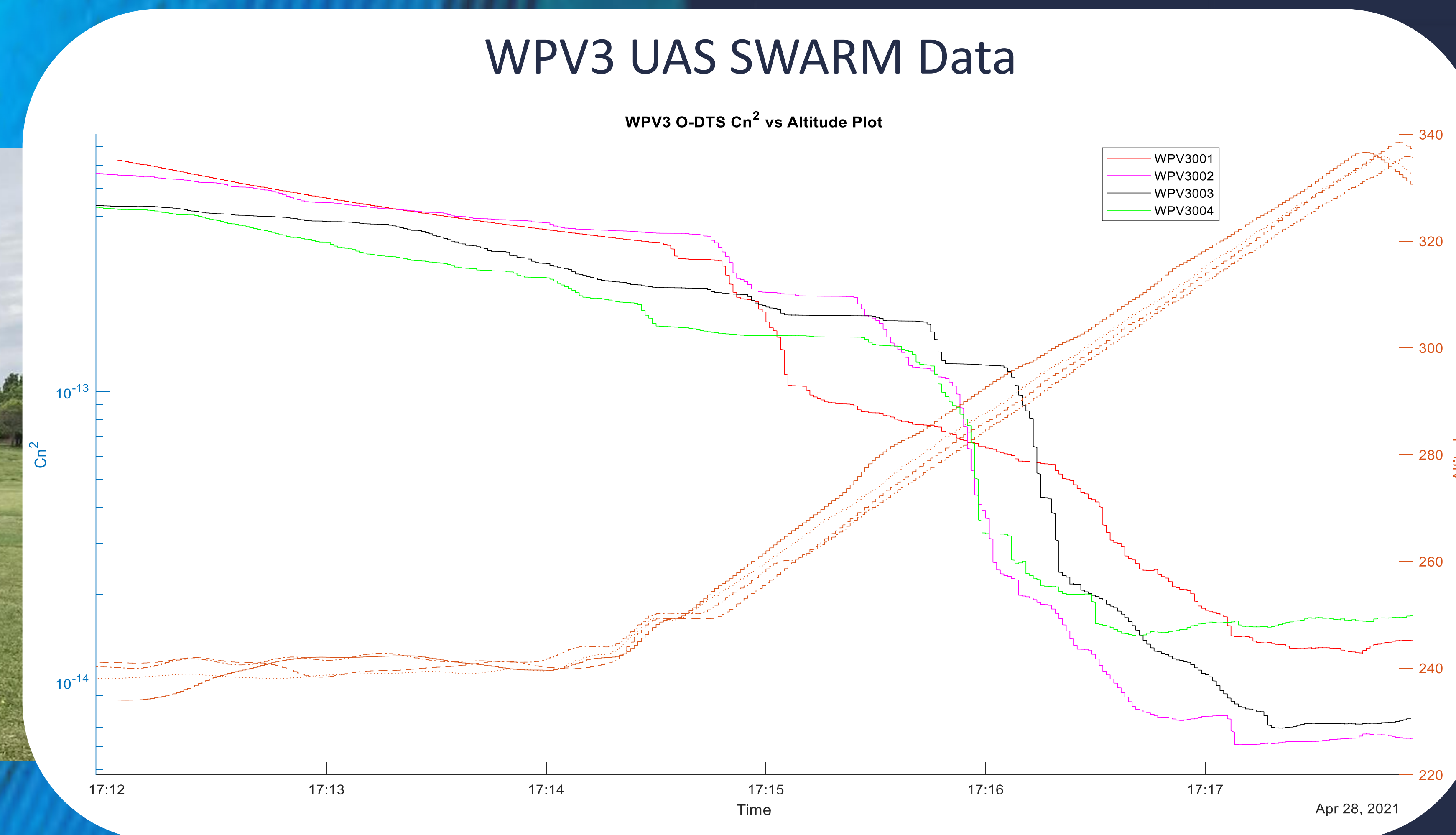


## UAS Payload Base Specifications

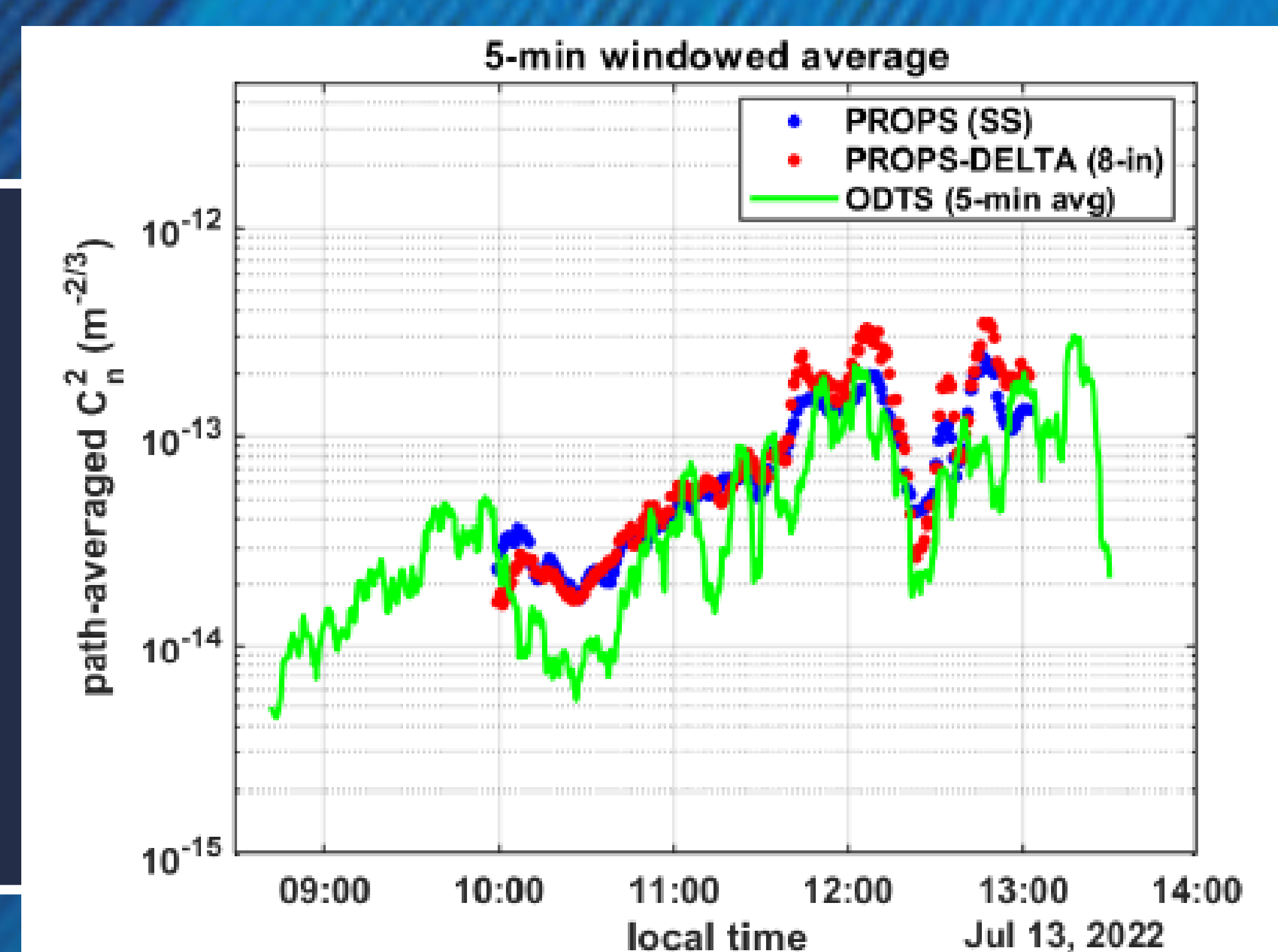
- Trisonica Mini Sonic Anemometer
  - o Temperature
  - o Pressure
  - o Relative Humidity
  - o 3D wind Speed
- Barometric Pressure Altimeter
- GPS
- LCD Data Display
- Telemetry Data Link
- SD Card Storage
- 35Hz Sample Rate
- Powered from UAS / 4-12S Battery
- Solar Pyranometer
- DTS Fine Wire Thermocouple
- Optical DTS – NIST traceable

| Atmospheric Characterization Payload |                     |               |      |  |
|--------------------------------------|---------------------|---------------|------|--|
| Component                            | Feature             | Specification | Unit |  |
| Wind Speed                           | Range               | 0-50          | m/s  |  |
|                                      | Resolution          | 0.1           | m/s  |  |
|                                      | Accuracy (0-15 m/s) | +/-0.1        | m/s  |  |
| Wind Direction                       | Accuracy (15-30m/s) | +/-2          | %    |  |
|                                      | Range (x,y)         | 360           | Deg  |  |
|                                      | Range (z)           | +/-30         | Deg  |  |
| Temperature                          | Resolution          | +/-1          | Deg  |  |
|                                      | Accuracy            | +/-1          | Deg  |  |
|                                      | Range               | -25-80        | C    |  |
| Humidity                             | Resolution          | 0.1           | C    |  |
|                                      | Accuracy            | +/-2          | C    |  |
| Pressure                             | Range               | 0-100         | %    |  |
|                                      | Accuracy            | +/-5          | %    |  |
| Tilt                                 | Range               | 50-115        | kPa  |  |
|                                      | Accuracy            | +/-1          | kPa  |  |
| Compass                              | Pitch and Roll      | +/-180        | Deg  |  |
|                                      | Accuracy            | +/-0.5        | Deg  |  |
| Data Storage                         | Heading             | 360           | Deg  |  |
|                                      | Accuracy            | +/-5          | Deg  |  |
| System Speed                         | SD Card             | 32            | GB   |  |
|                                      | Frequency           | 20            | Hz   |  |
| <b>Total Weight:</b>                 |                     | 0.75          | kg   |  |
| <b>Boom Height:</b>                  |                     | 46            | cm   |  |
| <b>Input Power:</b>                  |                     | 14-50         | VDC  |  |

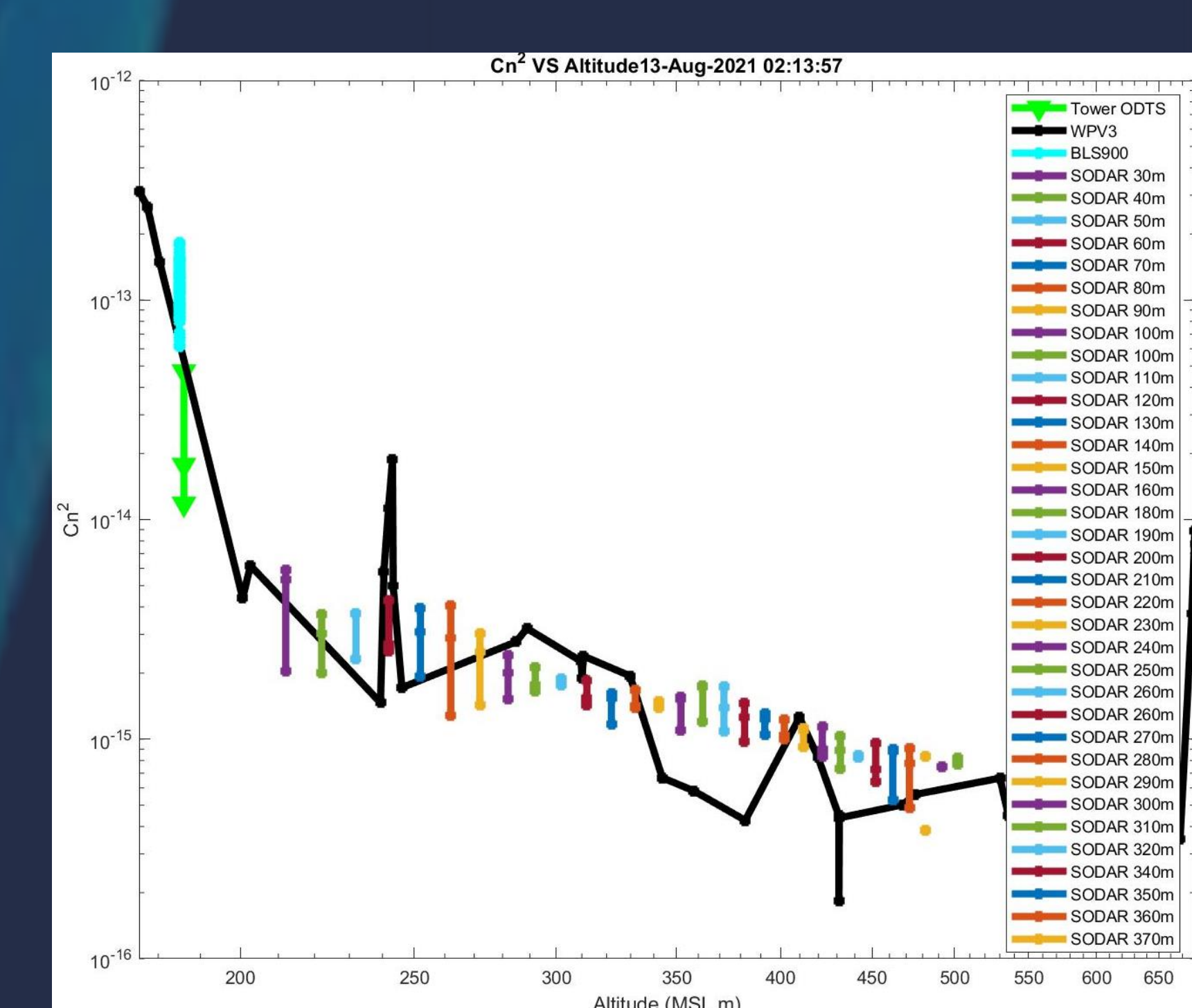
## UAS Weather Swarm



## WPV3 ODS vs DELTA



## SODAR Data



## Integrated Path Test

- July 12-13<sup>th</sup> 2022 – BlueHalo and MZA collected atmospheric turbulence data in combined effort
- A WPV3 ODS was mounted to a UAS to collect data while MZA's DELTA device was placed on ground
- A target board was mounted on the UAS for the DELTA to observe – this provided an integrated path of  $C_n^2$  values as opposed to the previous nodal tests
- The maximum distance measured was 750 meters from DELTA to target board (UAS mounted) – maximum altitude of the UAS was 100 meters



100 m